

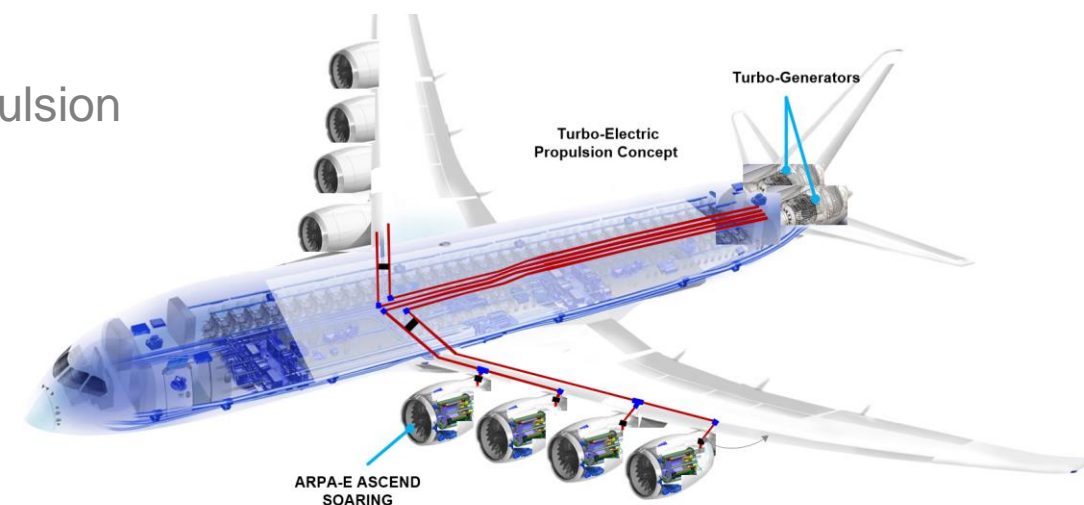
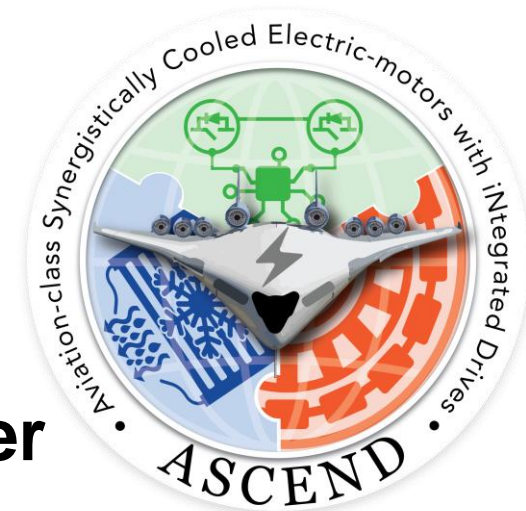
Superconducting Motor and Cryo-Cooled Inverter Engine: SOARING

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Raytheon Technologies Research Center

Project Vision

Disrupting Multi-MW Class Aircraft Propulsion
through extreme power density:

- Fully Superconducting AC motor
- Cryocooled motor drive and
- Adaptive Magnetocaloric Cryo-cooler



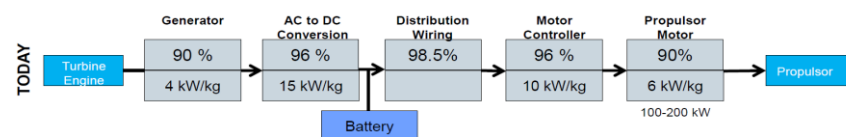
REEACH / ASCEND / CABLES
Annual Program Review Meeting
June 28-30, 2022



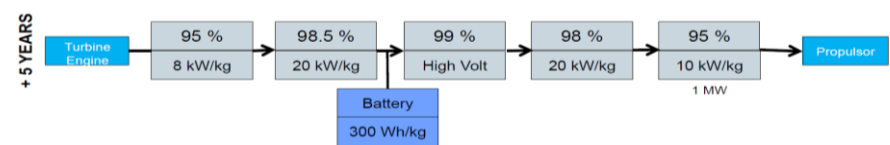
Background: 20MW Electrified Aircraft

Electric Propulsion System

SOA Today $Sp = 1.5 \text{ kW/kg}$, $\eta_{elec} = 73\%$



+ 5 Years $Sp = 2.8 \text{ kW/kg}$, $\eta_{elec} = 86\%$

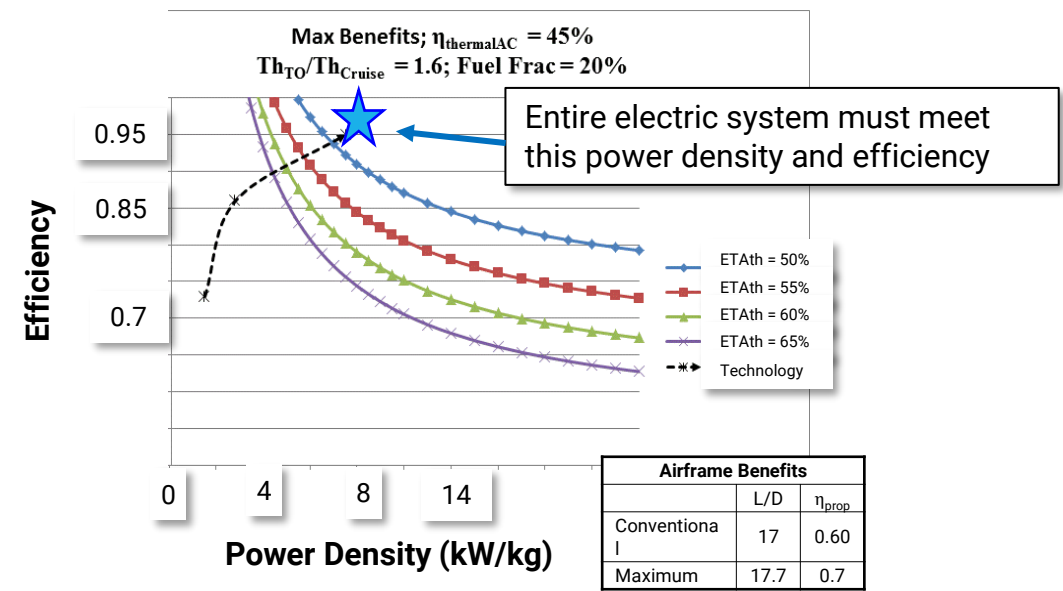
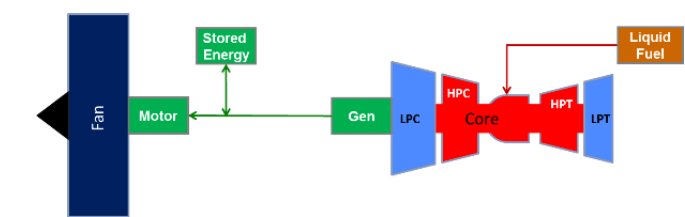


+10 Years Power Pillar $Sp = 7.5 \text{ kW/kg}$, $\eta_{elec} = 95\%$



Power density of components: X3 – X4 over SOA Today

20MW Series Turbo-electric

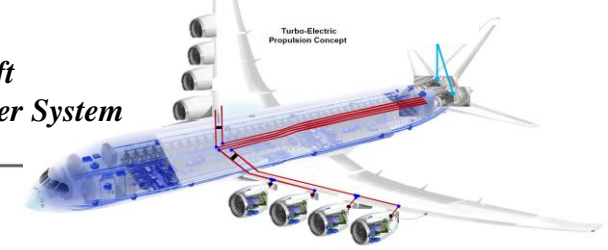


- Electric propulsion technology to achieve fuel savings for given configuration
 - Electric System Efficiency and Specific Power
 - Breakeven curves: fuel weight reduction equals weight of electric drive system
 - Above curves, net benefit

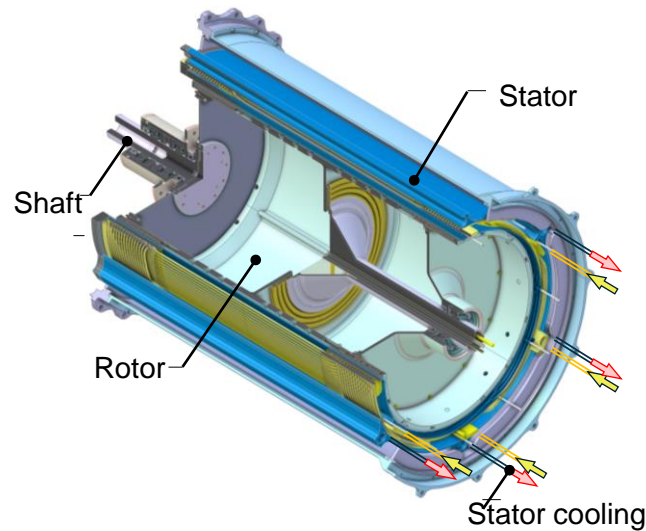
P. Kshirsagar et.al. AIAA 2020, Adapted from Jansen et al. 2015, 2017.

SOARING Overview

20MW Conceptual Aircraft
Cryofuel (Bio-LNG) Enabled Power System

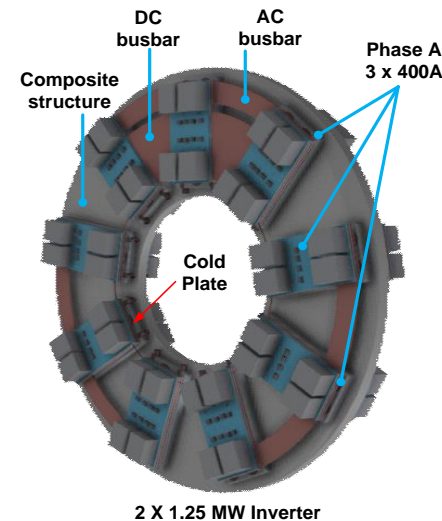


Fully Superconducting Motor (ACSYM) Direct Drive (no-gear)



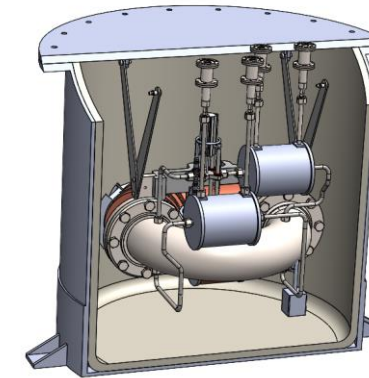
2.5MW, 5000rpm, 20K
Efficiency > 99%

Motor Drive (CROWN)



2.5MW, 120K
Efficiency > 99%

Magnetocaloric cryocooler (AMAC)



120K to 20K (AMAC)
COP target 0.6 (3x over SOA),
Power density Target 4W/kg (3x to 5x over SOA)

Phase-1: Conceptual Design

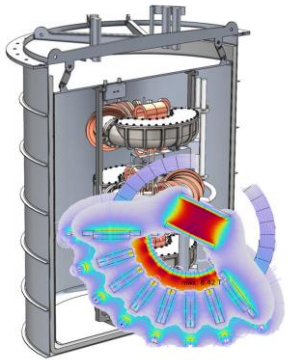
Phase-1: 40W Demonstration

SOARING: Team Organization

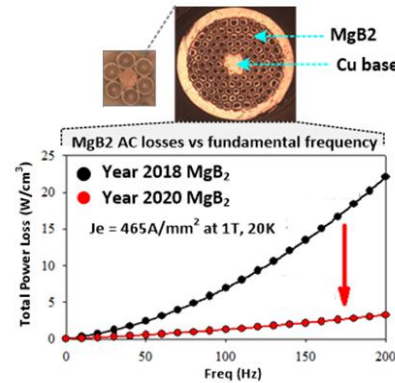
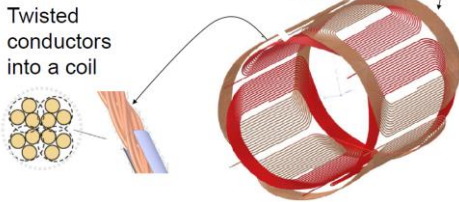
Phase-1



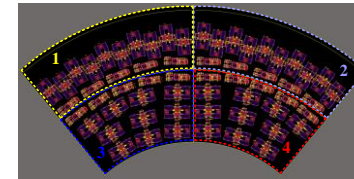
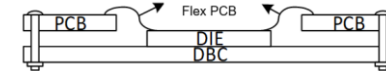
EMVIONIC



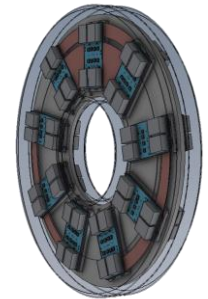
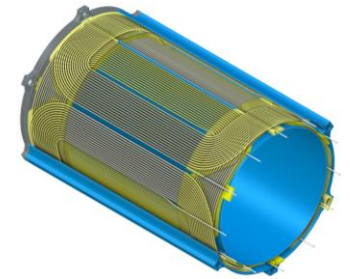
80K to 20K 40W
AMAC Demo



Concept design of superconducting
wires for stator and rotor



Concept design package
of GaN based cryocooled
motor drive

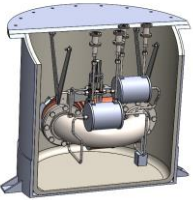


Concept design integration of
motor, motor-drive and AMAC

SOARING: TMS Performance +Motor + Power Electronics

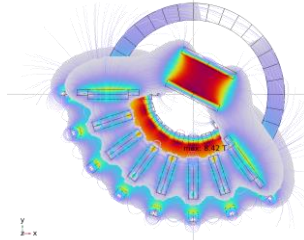
TMS-AMAC

Phase-1 Demo



*40 W 120K to 20K (AMAC)
COP target 0.6 (3x over SOA),
Power density 4W/kg (3x to 5x over SOA)*

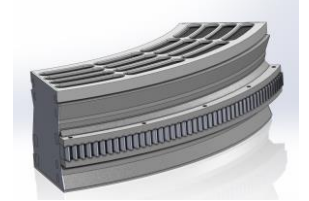
High Field Magnet Design



Rotary Seal Testing

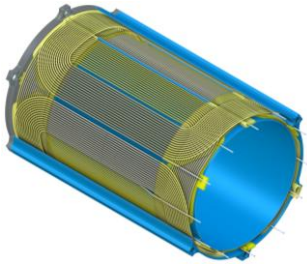


Rotating regenerator frame design



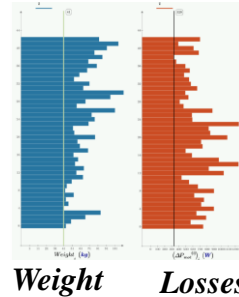
MOTOR

Phase-1 Concept Design

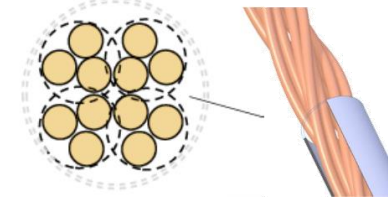


*Superconducting Motor
Take off Efficiency > 99%
Direct Drive (no-gearbox)*

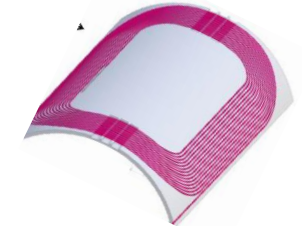
Design tradeoffs



AC Superconductors Manufacturing, packaging, cooling, and loss estimation



Wind and react manufacturing, direct cooling channels



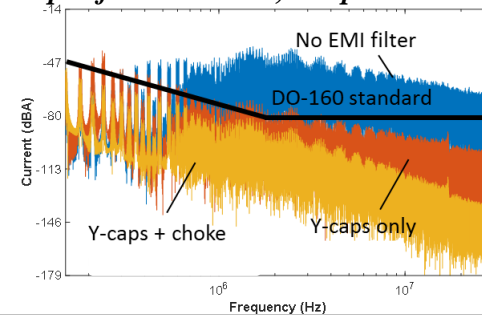
MOTOR DRIVE

Phase-1 Concept Design

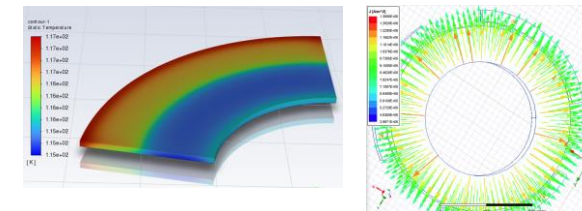


*Cryo Motor Drive
Takeoff Efficiency > 99%*

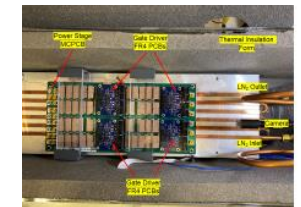
Input filter DO-160, Output THD < 1%



Power Module, Busbar, Capacitor Cooling

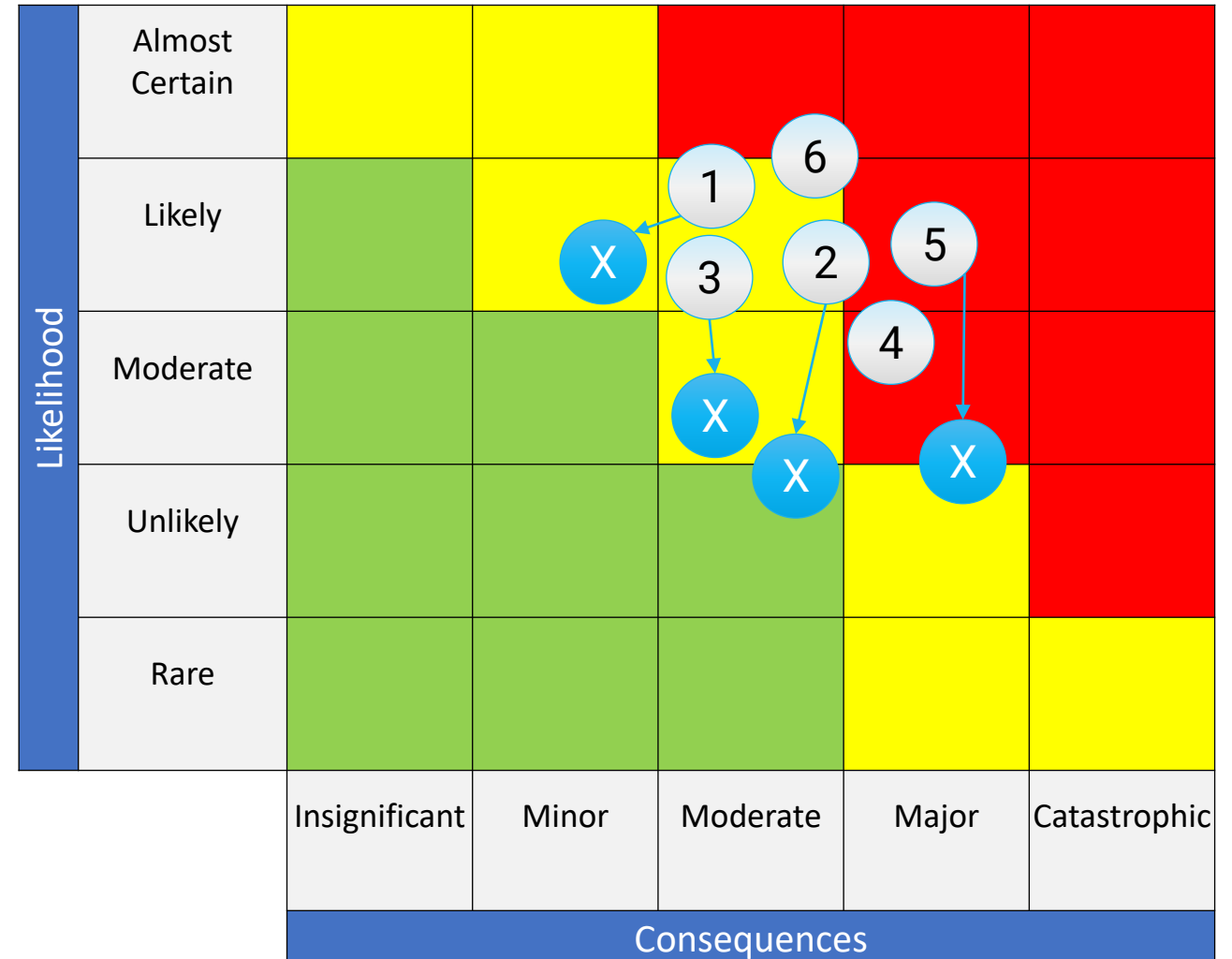


Low power Cryogenic converter testing



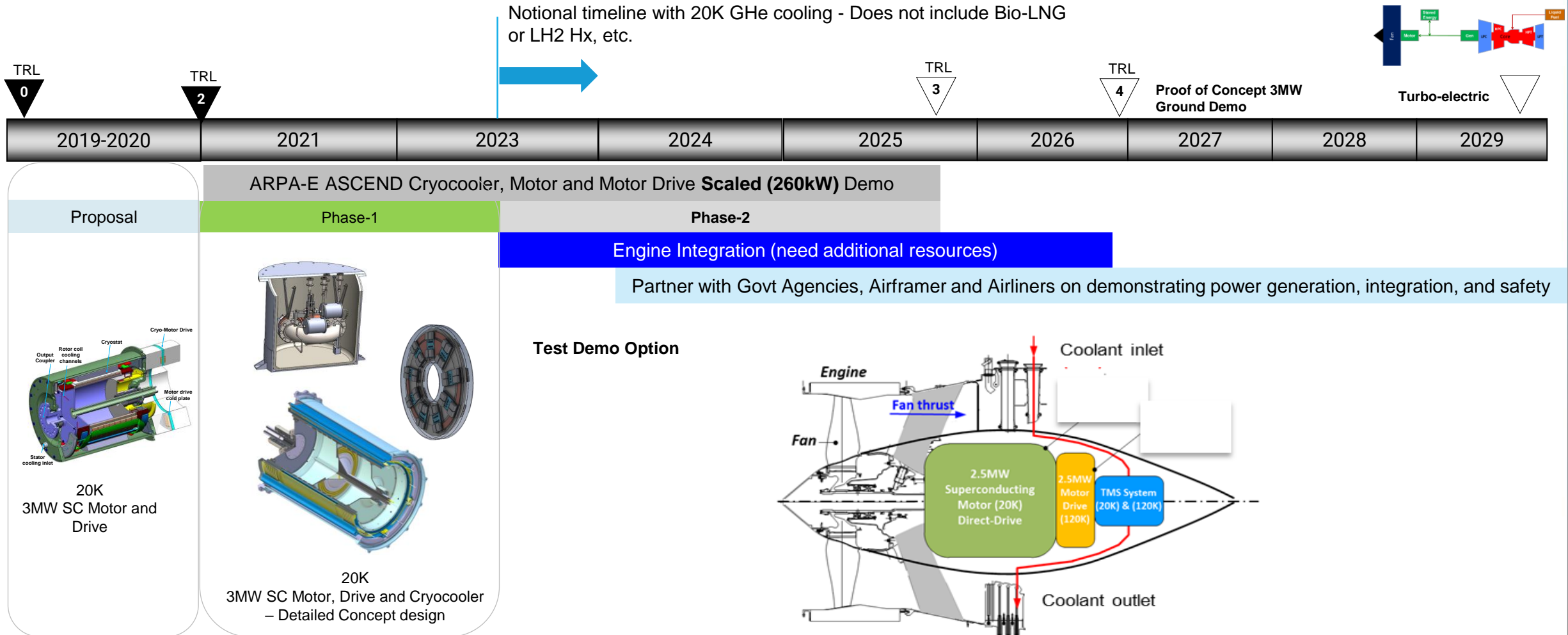
SOARING Risk Update

Risk	#	Trend
Manufacturing, support and cooling of the SC coils, AC Losses	1	↓
Effectiveness of the seals at cryo temperatures	2	↓
High lead losses between motor and drive	3	↓
Weight of AMAC	4	■
Low cost high strength DC magnets for AMAC	5	↓
Cost of the SOARING system	6	■



Technology-to-Market Approach

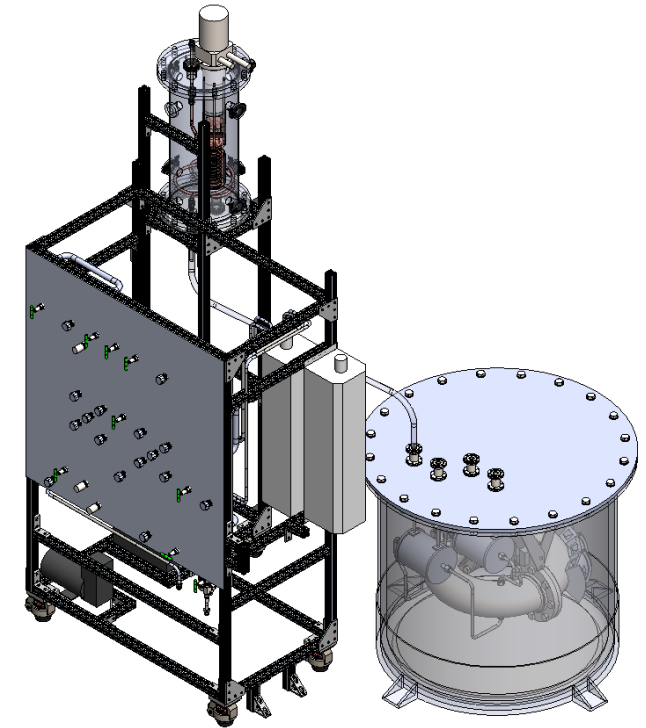
Superconducting motors scales better at higher power levels



Looking Ahead – What is anticipated for an Eventual Phase II?

- ▶ *Phase I*
 - *Motor: Proof of concept stator coil coupon*
 - *AMAC Demo*
- ▶ *Phase II.*
 - *Preliminary design of motor, motor drive and TMS*
 - *Detailed design of motor, motor drive and TMS*
 - *Sub-component integration and scaled demonstration*
- ▶ *T2M during the eventual Phase II.*
 - *Aerospace Applications: Collaboration with Propulsion partner, Airframer, and Airliner*
 - *Liquid hydrogen applications (Liquefier manufacturers)*
 - *Non-Aerospace Applications: Transportation Vehicles, and Space Applications*

AMAC Demonstration setup at PNNL



Q & A



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ENERGY

<https://arpa-e.energy.gov>